

**SIZE EFFECT OF AN ELLIPTIC INCLUSION IN ANTI-PLANE STRAIN COUPLE STRESS ELASTICITY**

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**1. Introduction**

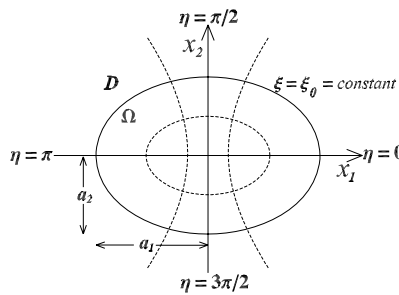
The application of higher order continuum theories, with size effect considerations, have recently been spread in the micro and nano-scale studies. One famous version of these theories, proposed by Mindlin[1], is the couple stress theory. This paper utilizes this theory to study the anti-plane problems of elliptic inclusions.

**2. Solution of the governing equations**

The governing field equation for the anti-plane problems of couple stress elasticity within a centrosymmetric isotropic material is given by

$$(1) \quad \nabla^2 u_3 - \ell^2 \nabla^4 u_3 = 0,$$

where  $\ell$  is the characteristic length and  $u_3$  is the out of plane displacement, [2]. Concerned with the problems of elliptic cylindrical inclusions, the solution of Eq.(1) is sought in elliptic coordinates,  $(\xi, \eta)$  with  $(x_1, x_2) = c_0(\cosh \xi \cos \eta, \sinh \xi \sin \eta)$ , Fig.1. Where  $c_0$  is a positive constant.



**Figure 1.** An elliptic domain within an infinite medium.

The general solution of this equation is taken as  $\bar{u}_3 + \bar{\bar{u}}_3$ , provided that

$$(2) \quad \frac{\partial^2}{\partial \xi^2} \bar{u}_3 + \frac{\partial^2}{\partial \eta^2} \bar{u}_3 = 0, \quad \frac{\partial^2}{\partial \xi^2} \bar{\bar{u}}_3 + \frac{\partial^2}{\partial \eta^2} \bar{\bar{u}}_3 - \frac{c_0^2}{2\ell^2} (\cosh 2\xi - \cos 2\eta) \bar{\bar{u}}_3 = 0.$$

Consider an elliptic domain,  $\Omega$  within an infinite medium,  $D$ , as shown in Fig.1. In the elliptic coordinate system the interface between  $\Omega$  and  $D$  is described by  $\xi = \xi_0$ . The long and short semi-axes of  $\Omega$  are denoted by  $a_1$  and  $a_2$ , respectively. The general solution of Eq.(1), periodic in  $\eta$ , associated with the exterior and interior points of  $\Omega$  are respectively given by  $u_3^{(1)}$  and  $u_3^{(2)}$  as:

$$(3) \quad u_3^{(1)}(\xi, \eta) = \sum_{k=1}^{\infty} (\bar{a}_{k1} \cos k\eta + \bar{b}_{k1} \sin k\eta) (\cosh k\xi - \sinh k\xi) + \bar{\bar{a}}_{01} c e_0(q, \eta) K e_0(q, \xi) + \sum_{k=1}^{\infty} \bar{\bar{a}}_{k1} c e_k(q, \eta) K e_k(q, \xi) + \bar{\bar{b}}_{k1} s e_k(q, \eta) K o_k(q, \xi),$$

and

$$(4) \quad u_3^{(2)}(\xi, \eta) = \sum_{k=1}^{\infty} (\bar{a}_{k2} \cos k\eta \cosh k\xi + \bar{b}_{k2} \sin k\eta \sinh k\xi) + \\ + \bar{a}_{02} c e_0(q, \eta) I e_0(q, \xi) + \sum_{k=1}^{\infty} \bar{a}_{k2} c e_k(q, \eta) I e_k(q, \xi) + \bar{b}_{k2} s e_k(q, \eta) I o_k(q, \xi),$$

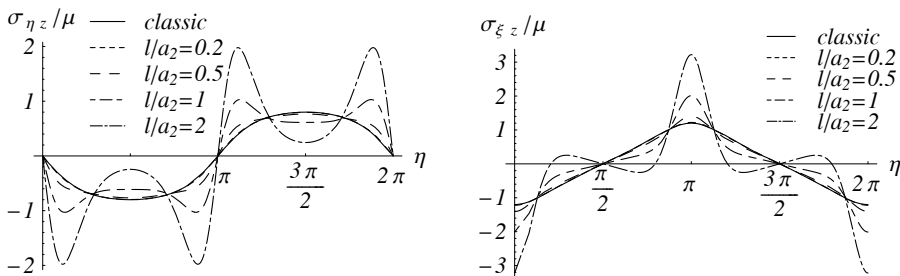
where  $q = -c_0^2/4\ell^2$ . Here  $c e_k$  and  $s e_k$  are the angular Mathieu functions and  $K e_k$ ,  $I e_k$ ,  $K o_k$  and  $I o_k$  are the radial Bessel type Mathieu functions. Assume that the displacement field  $u_3^* = 2e_{3i}^* x_i + e_{3ij}^* x_i x_j + \dots$  is given inside  $\Omega$ , where the summation is performed on  $i, j = 1, 2$  and  $e_{3i}^*, e_{3ij}^*, \dots$  stand for the eigenstrains [3]. The unknown coefficients in Eqs.(3-4) are determined through satisfaction of the following conditions on  $\xi = \xi_0$ ,

$$(5) \quad \bar{T}_3^{(1)} = \bar{T}_3^{(2)}, \quad \bar{M}_\eta^{(1)} = \bar{M}_\eta^{(2)}, \quad u_3^{(1)} = u_3^{(2)} + u_3^*, \quad \sigma_{z\xi}^{(1)} = \sigma_{z\xi}^{(2)},$$

where  $\bar{T}_3$  and  $\bar{M}_\eta$  are the reduced traction components and  $\sigma_{ij}$  is the component of the stress tensor. The superscripts (1) and (2) over a field quantity implies that it is derived from the displacements,  $u_3^{(1)}$  and  $u_3^{(2)}$ , respectively.

### 3. Numerical results and conclusion

Suppose  $\Omega$  is an inclusion with uniform eigenstrain,  $e_{13}^* = 1$ .  $\Omega$  and  $D$  are made of same material, and so they have the same shear modulus,  $\mu$  and characteristic length,  $\ell$ . To examine the size effect, various ratios for  $\frac{\ell}{a_2}$  are considered. In a special case with  $a_1 = a_2$ , the results via the present formulation reduce to the results derived from the work of Lubarda [2]. For a case where  $a_1 = 1.5a_2$ , the shear stresses  $\sigma_{\eta z}$  and  $\sigma_{\xi z}$  just outside of the inclusion along the inclusion-matrix interface are shown in Fig.2. This figure verifies that the results of the present study approach the classical solutions as the inclusion dimensions grow. It is observed that  $\sigma_{\xi z}$  attains its maximum at the end points of the long axis of the inclusion, while  $\sigma_{\eta z}$  vanishes at these points.



**Figure 2.** The stress distribution along the inclusion-matrix interface, approached from the matrix.

### 4. References

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