

BRIEF NOTES

Governing equations for simple continuum feathers

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WE ADOPT a special Cosserat material curve as a continuum model for simple feather structures. We then present the governing equations without presenting the details of derivation. Finally, a set of linear equations for a simple feather structure is presented.

1. Introductions

RECENT technological trends towards mechanisms duplicating the flight of birds and insects [1] demand dynamical analyses of feather structures. This is required for deter-

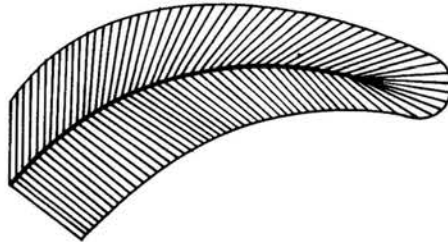


FIG. 1. A simple feather structure.

mining the configurational stability of such structures in flight. For some expositions regarding such analyses the reader is referred to [2]–[10].

The aim of the present work is to adopt a special class of Cosserat material curve as a continuum model for a simple feather structure (Fig. 1). We shall not present the derivation of the governing equations in detail since this can be easily found in [12]–[15].

2. Kinematics of a simple feather structure

We consider a simple feather structure a portion of which is shown in Fig. 1. We adopt a special class of Cosserat material curve for our model such that the generalized coordinate $\mathbf{P} \equiv (\mathbf{r}, \mathbf{d}_1, \mathbf{d}_2)$ of any material point X on the curve is an ordered set subject to the following kinematical constraints:

$$(2.1) \quad \mathbf{d} = \hat{R}(X, t)\mathbf{d}_2,$$

$$(2.2) \quad \mathbf{d}_2 = \hat{R}^T(X, t)\mathbf{d}_1,$$

$$(2.3) \quad \hat{R}^{-1} = \hat{R}^T, \quad \det \hat{R} \neq 0,$$

where $\mathbf{r}(X, t)$ is the position vector of each material point X at time t , $d_i(X, t)$, $i = 1, 2$ are two deformable vectors (directors) attached to each point, X_0 playing the role of pairs of feather flags, \hat{R} is an orthogonal and nonsingular rotation matrix, and the superscript T denotes the transpose. Note that the family of vectors $\mathbf{P} \equiv (\mathbf{r}, \mathbf{d}, \mathbf{d}_2)$ generate a nine-dimensional vector space \mathcal{E} whose vector algebra is discussed by SHAHINPOOR [14, 15]. If \mathcal{K} is an operator in \mathcal{E} defined in such a manner that

$$(2.4) \quad \mathcal{K}: \mathcal{E} \rightarrow \mathcal{E},$$

$$(2.5) \quad \mathbf{Q} \equiv \mathcal{K}\mathbf{P} \equiv (\varrho_0\mathbf{r} + \varrho_{01}\mathbf{d}_1 + \varrho_{02}\mathbf{d}_2, \varrho_{01}\mathbf{r}_1 + \varrho_{12}\mathbf{d}_2 + \varrho_1\mathbf{d}_1, \varrho_2\mathbf{d}_2 + \varrho_{12}\mathbf{d}_1 + \varrho_{02}\mathbf{r}),$$

where $\varrho_0, \varrho_{01}, \varrho_{02}, \varrho_{1,2}, \varrho_1, \varrho_2$ are all material line measures and functions of X alone. We can now easily derive the governing thermodynamical equations as

$$(2.6) \quad \frac{\partial \mathcal{K}}{\partial t} + \nabla \cdot (\dot{\mathbf{r}}\mathcal{K}) = 0,$$

$$(2.7) \quad \mathcal{K}\ddot{\mathbf{P}} = \mathbf{T}_{,X} + \mathcal{K}\mathbf{F} - \mathbf{M},$$

$$(2.8) \quad \mathbf{P}_X \hat{\times} \mathbf{F} + \mathbf{P} \hat{\times} \mathbf{M} = 0,$$

$$(2.9) \quad \dot{e} = \mathbf{T} \circ \dot{\mathbf{P}}_{,X} + \dot{\mathbf{P}} \circ \mathbf{M} + r + q_{,X},$$

$$(2.10) \quad \dot{\theta}\eta - r - q_{,X} - \theta^{-1}q\theta_{,X} \geq 0.$$

If one introduces a Helmholtz free energy function $\psi = e - \theta\eta$, then Eqs. (2.9) and (2.10) can be combined to yield the Clausius-Duhem inequality:

$$(2.11) \quad -\dot{\psi} - \dot{\theta}\eta + \mathbf{T} \circ \dot{\mathbf{P}}_{,X} + \dot{\mathbf{P}} \circ \mathbf{M} + \theta^{-1}q\theta_{,X} \geq 0.$$

Constitutive relations

We let $\psi, \eta, \mathbf{T}, \mathbf{M}, q$ be the dependent constitutive functions. Taking Truesdell's "equi-presence principle" [16] we can write

$$(2.12) \quad \psi = \hat{\psi}(\mathbf{P}, \mathbf{P}_{,X}, \theta, \dot{\theta}, \theta_{,X}; X),$$

$$(2.13) \quad \eta = \hat{\eta}(\mathbf{P}, \mathbf{P}_{,X}, \theta, \dot{\theta}, \theta_{,X}; X),$$

$$(2.14) \quad \mathbf{T} = \hat{\mathbf{T}}(\mathbf{P}, \mathbf{P}_{,X}, \theta, \dot{\theta}, \theta_{,X}; X),$$

$$(2.15) \quad \mathbf{M} = \hat{\mathbf{M}}(\mathbf{P}, \mathbf{P}_{,X}, \theta, \dot{\theta}, \theta_{,X}; X),$$

$$(2.16) \quad q = \hat{q}(\mathbf{P}, \mathbf{P}_{,X}, \theta, \dot{\theta}, \theta_{,X}; X).$$

Note that no dependence on $\dot{\mathbf{P}}$ is assumed because the axiom of objectivity eliminates such constitutive dependences. With the above descriptions the inequality (2.11) expands to

$$(2.17) \quad \left(\mathbf{T} - \frac{\partial \psi}{\partial \mathbf{P}_{,X}} \right) \circ \dot{\mathbf{P}} + \left(\mathbf{M} - \frac{\partial \psi}{\partial \mathbf{P}} \right) \circ \dot{\mathbf{P}} \\ - \left(\eta + \frac{\partial \psi}{\partial \theta} \right) \dot{\theta} - \left(\frac{\partial \psi}{\partial \dot{\theta}} \right) \ddot{\theta} - \left(\frac{\partial \psi}{\partial \theta_{,X}} \right) \dot{\theta}_{,X} + (\theta q^{-1})\theta_{,X} \geq 0.$$

We require that the inequality (2.17) hold for all arbitrary variations in $\dot{\mathbf{P}}_{,x}$, $\dot{\mathbf{P}}$, $\hat{\mathbf{P}}$, $\dot{\theta}$, $\dot{\theta}_{,x}$. We then find that

$$(2.18) \quad \mathbf{T} = \frac{\partial \psi}{\partial \mathbf{P}_{,x}}, \quad \frac{\partial \psi}{\partial \dot{\mathbf{P}}} = \frac{\partial \psi}{\partial \dot{\theta}} = \frac{\partial \psi}{\partial \dot{\theta}_{,x}} = 0, \quad M = \frac{\partial \psi}{\partial \mathbf{P}},$$

$$(2.19) \quad -\left(\eta + \frac{\partial \psi}{\partial \dot{\theta}}\right) \dot{\theta} + \theta^{-1} q \theta_{,x} \geq 0.$$

If we assume that thermodynamic equilibrium is achieved where the temperature is uniform throughout the body and is constant in time, i.e. $\dot{\theta} = \theta_{,x} = 0$, for all X and t , then it immediately follows from Eqs. (2.18)₂ and the inequality (2.19) that

$$(2.20) \quad \eta|_E = -\frac{\partial \psi}{\partial \theta}|_E, \quad q|_E = 0,$$

$$(2.21) \quad -2 \frac{\partial \eta}{\partial \dot{\theta}}|_E \geq 0,$$

$$(2.22) \quad -\frac{\partial \eta}{\partial \theta_{,x}}|_E + \theta^{-1} \frac{\partial q}{\partial \dot{\theta}}|_E \geq 0,$$

$$(2.23) \quad \theta^{-1} \frac{\partial q}{\partial \theta_{,x}}|_E \geq 0,$$

where $|_E$ implies at equilibrium.

Similarly, we can define a thermodynamical steady state situation such that $\dot{\theta} = 0$ for all X and t . Thus we are led to the inequality

$$(2.24) \quad \theta^{-1} q|_{ss} \theta_{,x} \geq 0.$$

One may use more sophisticated entropy principles (MÜLLER [17], GREEN and LAWS [18]) and arrive at slightly different constitutive relations.

The constitutive functions finally take the form

$$(2.25) \quad \psi = \hat{\psi}(\mathbf{P}, \mathbf{P}_{,x}, \theta; X),$$

$$(2.26) \quad \mathbf{T} = \hat{\mathbf{T}}(\mathbf{P}, \mathbf{P}_{,x}, \theta; X),$$

$$(2.27) \quad \mathbf{M} = \hat{\mathbf{M}}(\mathbf{P}, \mathbf{P}_{,x}, \theta; X),$$

$$(2.28) \quad \eta = \hat{\eta}(\mathbf{P}, \mathbf{P}_{,x}, \dot{\theta}, \theta_{,x}, \theta; X),$$

$$(2.29) \quad q = \hat{q}(\mathbf{P}, \mathbf{P}_{,x}, \dot{\theta}, \theta_{,x}, \theta; X).$$

Note that if

$$(2.30) \quad f = \hat{f}(\mathbf{P}, \mathbf{P}_{,x}, \theta, \theta_{,x}, \dot{\theta}; X),$$

$$(2.31) \quad g = \hat{g}(P, \theta_{,x}, \theta, P \theta_{,x}, \dot{\theta}; X),$$

are arbitrary twice differentiable functions satisfying

$$(2.32) \quad -f \dot{\theta} + g \theta_{,x} \geq 0,$$

then we can write from the inequalities (2.19) and (2.32)

$$(2.33) \quad \eta = \frac{-\partial\psi}{\partial\theta} + f - a_1\dot{\theta} + a_2\theta_{,x},$$

$$(2.34) \quad \theta^{-1}q = g + b_1\dot{\theta} + b_2\theta_{,x},$$

$$(2.35) \quad a_1 \geq 0, \quad b_2 \geq 0, \quad a_2 + b_1 \geq 0.$$

If there exists a differential constraint of the form

$$(2.36) \quad \Phi_1 \circ \mathbf{P}_{,x} + \Phi_2 \circ \mathbf{P} + \Phi_3 \dot{\theta} + \Phi_4 \hat{\theta} + \Phi_5 \dot{\theta}_{,x} = 0,$$

then

$$(2.37) \quad \mathbf{T} = \frac{\partial\psi}{\partial\mathbf{P}_{,x}} + \lambda\Phi,$$

$$(2.38) \quad \mathbf{M} = \frac{\partial\psi}{\partial\mathbf{P}} + \lambda\Phi_2,$$

$$(2.39) \quad -\left(\eta + \frac{\partial\psi}{\partial\theta} + \lambda\Phi_3\right)\dot{\theta} + (\theta^{-1}q\theta_{,x}) \geq 0,$$

$$(2.40) \quad \frac{\partial\psi}{\partial\mathbf{P}} = 0, \quad \frac{\partial\psi}{\partial\dot{\theta}} - \lambda\Phi_4 = 0, \quad \frac{\partial\psi}{\partial\dot{\theta}_{,x}} - \lambda\Phi_5 = 0.$$

Finally, we present here equations corresponding to a linearized model:

$$\psi = \frac{1}{2}C_1 P_{,x}^2 + C_2 \tilde{P}^2 + C_3 \tilde{\mathbf{P}} \circ \mathbf{P}_{,x},$$

$$\mathbf{T} = C_1 \mathbf{P}_{,x} + C_3 \tilde{\mathbf{P}},$$

$$\mathbf{M} = C_3 \mathbf{P}_{,x} + C_2 \mathbf{P},$$

$$\eta = f - a_1 \dot{\theta} + a_2 \theta_{,x},$$

$$\theta^{-1}q = g + b_1 \dot{\theta} + b_2 \theta_{,x},$$

$$\mathcal{K}\ddot{\mathbf{P}} = C_1 \mathbf{P}_{,xx} + C_3 \tilde{\mathbf{P}}_{,x} + \mathcal{K}\mathbf{F} - C_3 \mathbf{P}_{,x} + C_2 \tilde{\mathbf{P}},$$

where $C_1, C_2, C_3, a_1, a_2, b_1, b_2$ are material constants.

References

1. S. CORSIN, Dept. Mats. Sci., The Johns Hopkins University July 1976, Private Communication.
2. E. J. MAREY, *La machine animale*, (Livre 3, me, Chap. 111-VI), Librairie G. Bailliere, Paris 1973.
3. J. H. STORER, *The flight of birds*, Cranbrook Int. Sci., Bloomfield Hills, Mich., 1948.
4. I. N. VINOGRADOV, *The aerodynamics of soaring bird flight*, Dosarm, Moscow 1951, Roy. Aircraft Estab., (Franborough), Library Transl. No. 846, Ministry of Aviation, London, Jan. 1960.
5. G. S. VASILEV, *Principles of flight of models with flapping wings*, Moscow 1953; Tech., Intell. Transl., Air Tech. Intell. Ctr., Wright-Patterson Air Force Base, Ohio 1954.
6. C. D. CONE, Jr., (Rept. So), *A mathematical analysis of the dynamic soaring flight of the albatross with ecological interpretations*, Special Scientific Report 50, Va., Inst. of Marine Sciences, Gloucester Point, Va., May 1964.

7. H. HERTEL, *Structure, form and movement*, Mainz 1903; Biology and Technology Series Reinhold Publ. Corp., N. Y., English Language Edition, 1966.
8. C. D. CONE, Jr., (Rept. 521) *The aerodynamics of flapping bird flight*, Special Scientific Rep., 52, Va., Inst. of Marine Sciences, Gloucester Point, Vo., Oct., 1968.
9. R. MC. ALEXANDER, *Animal mechanics*, Biology series, University of Washington Press, Seattle 1968.
10. D. URRY and K. URRY, *Flying birds*, harper and row, N. Y. 1969.
11. C. J. PENNYCUICK, *Animal flight*, Edward Arnold (Publ.), Ltd., London 1972.
12. S. S. ANTMAN, *The theory of rods*, Handbuch der Physik, Bd. VIa/2 (Edited by FLUGGE and TRUESDELL), Springer-Verlag, Berlin 1972.
13. M. SHAHINPOOR, *Plane waves and hadamard stability in generalized thin elastic rods*, Int. J. Solids Struct., **11**, 861-870, 1975.
14. M. SHAHINPOOR, *Hadamard stability of uniform helical structures*, J. Struct. Mech., **5**, 1, 1977.
15. M. SHAHINPOOR, *A continuum theory of simple feather structures*, Iran. J. Sci., Tech., **6**, 3, 1977.
16. C. TRUESDELL and W. NOLL, *The nonlinear field theories*, Ed. Flügge, Handbuch der Physik (Berlin—Springer—Verlag), 111/3, 1965.
17. I. MÜLLER, *On entropy production inequality*, Arch. Ration. Mech. Anal., **26**, 118-124, 1967.
18. A. E. GREEN and N. LAWS, *On the entropy production inequality*, Arch. Ration. Mech. Anal., **45**, 47, 1972.

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